General Theory of Vibrations of Cylindrical Tubes[†]

PART-III: UNCOUPLED FLEXURAL VIBRATIONS OF CLOSED TUBES

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Summary

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Flexural vibration characteristics of unstiffened doubly symmetric cylindrical tubes are studied using the theory proposed by the authors in Fart I. An exact solution of a simply supported tube with the cross-section given by $x = \frac{s^2}{4\pi^2} \sin \frac{2\pi s}{s}$ is presented. Natural frequencies of a simply supported tube of rectangular cross-section are obtained by using first and second order approximation equations. Numerical results indicate that it is essential to consider, at least, second order approximation equations to study the flexural vibration characteristics of doubly symmetric tubes.

ADDITIONAL NOTATION**

- Half the width of the tube

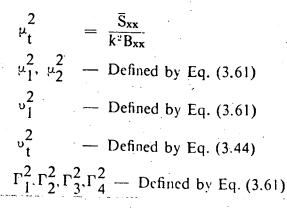
b — Half the depth of the tube $K_{t}^{4} = \frac{\omega^{2}\rho L^{4}A}{EB_{xx}}$ $k_{w}^{2} = \omega^{2}\rho L^{2}/E$ P = a/b Q = L/a $\alpha_{1}, \beta_{1} - Defined by Eq. (3.37)$ $\delta_{11} - Kronecker delta$ $\zeta_{1}, \zeta_{2} - Defined by Eq. (3.63)$ $\lambda_{t} - Defined by Eq. (3.63)$ $\lambda_{t}^{2}, \lambda_{2}^{2}, \lambda_{3}^{2}$ Roots of the polynomial (Eq. 3.39 and Eq. 3.65)

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**In addition to the notation used in Part I.

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3.0 Introduction

Flexural mode of vibration of doubly symmetric tubes, although uncoupled with torsion mode, generally involves warping motion also. This is because of the fact that the assumption of CSRMD allows complete freedom to warp. This is the typical difference between tube and classical beam theories. Warping displacement is a function of two variables; and so the analysis gets considerably complicated when compared to beam theory.

Over the years, this is one of the basic problems of intense research activity. Owing to the similarities in the formulation of beam and tube theories, several attempts were made to adopt or to modify the beam equations suitably. The necessary modifications are the incorporation of transverse shear, longitudinal inertia and shear lag effects. The differential equations are improved by including the rotary inertia^{2*} and transverse shear effects⁴. Investigations by subsequent workers⁵⁻¹⁰ confirmed that these led to more accurate determination of natural frequencies and mode shapes of tubular structures. Nevertheless, shear lag effects are not included into the differential equation in literature in an elegant way so far.

Modified beam theories cannot be as good for short tubes, since the shear lag effects have substantial influence on the natural frequencies and mode shapes. Budiansky and Kruszewski¹³ used Rayleigh-Ritz method to study the natural frequency characteristics of closed tubes. Mansfield²³ obtained a stress function solution for the stiffness of a rectangular box-section, utilising which the natural frequencies can be calculated. These investigations divulge the importance of shear lag effects in short tubes. However, it is difficult to adopt them to tubes of complex cross-sections.

The present theory involves the development of governing equations for the natural vibrations of thin cylindrical tubes of arbitrary cross-section, with the assumption of CSRMD. Using the Kantorovich form of Rayleigh-Ritz method, these are reduced to governing equations to various orders of approximation; an elegant splitting of warp results in the incorporation of shear lag effects into the governing differential equations. This work is reported in Ref. 32.

*References are given in Part I.

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In this report, we utilise the equations deduced in Part 1³² to study the flexural vibrational characteristics of two tubes.

(1) A simply-supported tube with the cross section

given by
$$x = \frac{S^2}{4\pi^2} \sin \frac{2\pi s}{S}$$
.

(2) A simply- supported rectangular tube.

The first has an exact solution and brings out the nature of frequency spectrum whereas the second does not have an exact solution, but shows that, at least, second order approximation equations need to be used in order to get reasonable estimates of natural frequencies.

3.1 Governing equations - rigorous formulation.

The governing equations of equilibrium are given by Eqs. (1.31)*

$$\frac{d^2 u}{dz^2} + k_u^2 u = -\frac{1}{S_{xx}} \frac{d}{dz} \oint \frac{\partial w}{\partial s} \frac{dx}{ds} t ds \qquad \dots \quad (3.1)$$

and the boundary conditions at each end are (Eqs. (1.30) and (1.35))

either
$$u = 0$$
 or $\frac{du}{dz} + \frac{1}{S_{xx}} \oint \frac{\partial w}{\partial s} \frac{dx}{ds}$ t $ds = 0$... (3.3)

either
$$w = 0$$
 or $\frac{\partial w}{\partial z} = 0$
and $\left(\frac{\partial w}{\partial s}\right)_{s=0} = \left(\frac{\partial w}{\partial s}\right)_{s=S}$... (3.4)

3.2 A simply supported tube with the boundary of the cross section given by $x = \frac{S^2}{4\pi^2} \sin \frac{2\pi s}{S}$ - exact solution

The boundary conditions in this case are

$$u(0) = u(1) = 0$$
 ... (3.5)

$$\frac{\partial w(0, s)}{\partial z} = \frac{\partial w(1, s)}{\partial z} = 0 \qquad \dots \qquad (3 6)$$

Examination of Eqs. (3.6) and (3.7) suggests the expression for w in the form

$$w = \sum_{m=1, 2, 3, ...}^{\infty} \sum_{n=1, 2, 3, ...}^{\infty} A_{mn} \cos m\pi z \sin \frac{2n\pi s}{s}$$
(3.8)

Equations referred to as (1...) can be seen in Part I.

August 1968] GENERAL THEORY OF VIBRATIONS OF CYLINDRICAL TUBES Substituting Eq. (3.8) in Eq. (3.1)

$$\frac{d^{2}u}{dz^{2}} + k_{u}^{2}u = \sum_{m=1, 2, 3, ...}^{\infty} \sum_{n=1, 2, 3, ...}^{\infty} A_{mn} \frac{2mn\pi^{2}}{SS_{xx}} \sin m\pi z$$

$$\oint \frac{dx}{ds} \cos \frac{2n\pi s}{S} ds \qquad ... (3.9)$$

The boundary of the cross section is given by

$$x = \frac{S^2}{4\pi^2} \sin \frac{2\pi s}{S} \qquad ... (3.10)$$

Substituting Eq. (3.10) in (3.9), we have

$$\frac{d^2 u}{dz^2} + k_u^2 u = \sum_{m=1, 2, 3, ...}^{\infty} \sum_{n=1, 2, 3, ...}^{\infty} A_{mn} \frac{mn\pi St}{2S_{xx}} \delta_{1n} \sin m\pi z$$
(3.11)

The value of S_{xx} is given by (Eq. (1.19))

$$S_{xx} = \oint \left(\frac{dx}{ds}\right)^2 t \, ds = \frac{S^3 t}{8\pi^2} \qquad \dots \qquad (3.12)$$

Substituting Eq. (3.12) in Eq. (3.11)

$$\frac{d^2 u}{dz^2} + k_u^2 u = \sum_{m=1, 2, 3, ...}^{\infty} \sum_{n=1, 2, 3, ...}^{\infty} A_{mn} \frac{4mn\pi^3}{S^2} \delta_{1n} \sin m\pi z$$
... (3.13)

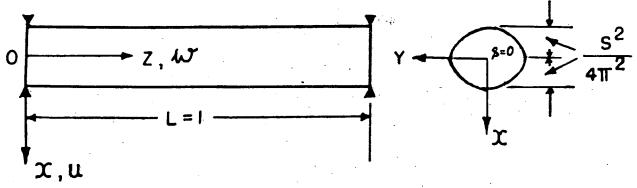
The solution of Eq. (3.13) is

$$u = A_{1} \sin k_{u}z + A_{2} \cos k_{u}z + \sum_{m=1, 2, 3, ...}^{\infty} \sum_{n=1, 2, 3, ...}^{\infty} \sum_{n=1, 2, 3, ...}^{\infty} \left\{ A_{mn} - \frac{4mn\pi^{3}\delta_{1n}}{S^{2}(-m^{2}\pi^{2} + k_{u}^{2})} \right\} \sin m\pi z \qquad ... \quad (3.14)$$

Satisfaction of the boundary conditions (3.5) yields $A_1 = A_2 = 0$... (3.15) Hence

$$u = \sum_{m=1, 2, 3, \dots}^{\infty} \sum_{n=1, 2, 3, \dots}^{\infty} A_{mn} \frac{4mn\pi^{3} \delta_{1n}}{s^{2} (-m^{2} \pi^{2} + k_{u}^{2})^{sin} m\pi z} \dots (3 6)$$





Fig, 3,1: A simply supported Tube with the Boundary of the Cross Section

$$given by x = \frac{S^2}{4\pi^2} \sin \frac{2\pi s}{s}$$

$$\sum_{m=1, 2, 3, ...}^{\infty} \sum_{n=1, 2, 3, ...}^{\infty} A_{mn} \left\{ -k^2 m^2 \pi^2 - \frac{4n^2 \pi^2}{S^2} + k_s^2 \right\}$$

$$= \sin \frac{2\pi s}{s} \sum_{m=1, 2, 3, ...}^{\infty} \sum_{n=1, 2, 3, ...}^{\infty} \left\{ A_{mn} \frac{4m^2 n \pi^4 \delta_{in}}{s^2 (-m^2 \pi^2 + k_u^2)} \right\} \cos m\pi z$$

$$(3.17)$$

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from which it follows

For the nontrivial solution of A_{mn} 's, we have the characteristic equation as

$$\left(-k^{2} m^{2} \pi^{2} - \frac{4n^{2} \pi^{2}}{S^{2}} + k^{2}_{S}\right) - \frac{4m^{2} n \pi^{4} \delta_{1n}}{S^{2} \left(-m^{2} \pi^{2} + k^{2}_{u}\right)} = 0$$

n, m = 1, 2, 3,... (3.19)

and the corresponding mode shapes are given by

$$w = A_{mn} \cos m \pi z \sin \frac{2n \pi s}{s}$$
 ... (3.20)
 $4mn \pi^3 \delta_{1n}$ sin mer

$$u = A_{\min} \frac{4 \min \pi^3}{S^2 \left(-m^2 \pi^2 + k_u^2\right)} \sin m\pi z$$

The case of $n \neq 1$ represents pure warping modes. Here natural frequencies are given by

$$k_{s}^{2} = k^{2}m^{2} \pi^{2} + \frac{4n^{2} \pi^{2}}{S^{2}} \qquad m = 1, 2, 3, ... \qquad ... \quad (3.21)$$

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and the mode shapes are

$$w = A_{mn} \cos m\pi z \sin \frac{2n \pi s}{s}$$

$$u = 0$$

$$m = 1, 2, 3, ...$$
(3.22)

and

$$m = 1, 2, 3, ...$$
 (3.2)

when n = 1, the characteristic equation is

$$\left(-k^{2}m^{2}\pi^{2}-\frac{4\pi^{2}}{S^{2}}+k_{s}^{2}\right)-\frac{4m^{2}\pi^{4}}{S^{2}\left(-m^{2}\pi^{2}+k_{u}^{2}\right)}=0 \qquad \dots (3.23)$$

$$m=1,2,3,\dots$$

and the mode shapes are

$$w = A_{m_1} \cos m\pi z \sin \frac{2\pi S}{S}$$
 ... (3.24)

$$= A_{m_1} \frac{4m \pi^3}{S^2 \left(-m^2 \pi^2 + k_u^2\right)} \sin m\pi z \qquad m = 1, 2, 3, ...$$

and

$$\frac{u_{\text{max}}}{w_{\text{max}}} = \frac{4m \pi^3}{S^2 \left(-m \pi^2 + k_u^2\right)} \qquad \dots (3.25)$$

Eqs. (3.23) and (3.25) are rewritten for convenience as

$$\left(-1+k_{u}^{*2}\right)\left(-1-\frac{\mu_{1}^{2}}{m^{2}\pi}+k_{w}^{*2}\right)-\frac{\mu_{1}^{2}}{m^{2}\pi^{2}}=0$$
 ... (3.26)

and

$$\frac{u_{\text{max}}}{w_{\text{max}}} = \frac{4\pi}{\text{mS}^2 \left(-1 + k_{\text{u}}^{*2}\right)} \qquad \dots \quad (3.27)$$

where

$$k_{u}^{*2} = k_{u}^{2} / m^{2} \pi^{2}$$

$$k_{w}^{*2} = k_{s}^{2} / k^{2} m^{2} \pi^{2} \dots (3 28)$$

$$\mu_{t}^{2} = 4 \pi^{2} / k^{2} S^{2}$$

Neglecting longitudinal inertia Eq. (3.26) reduces to a very simple form as

$$k_{u}^{*2} = \frac{m^{2} \pi^{2}}{\mu_{t}^{2} + m^{2} \pi^{2}} \qquad \dots \qquad (3.29)$$

Numerical results

The values of the frequency parameter k_u^{*2} associated with primarily transverse motion and the mode shapes, obtained by including longitudinal inertia (Eqs. 3.26

and 3.27) and by neglecting logitudinal inertia (Eqs. 3.29 and 3.27) are compared below. In these calculations S = 1.

Table 3.1 shows that the influence of longitudinal inertia, in this case is extremely small. As mentioned earlier inclusion of longitudinal inertia gives two infinite sets of frequencies, one of which has primarily transverse motion and the other involves primarily warping motion. The second set involving primarily warping motion does not appear if longitudinal inertia is neglected. As such primarily flexural modes only are compared in Table 3.1. Table 3.2 shows both the sets of frequencies and amplitude ratios obtained by using Eqs. (3 26) and (3.27) respectively, for the value S = 1.

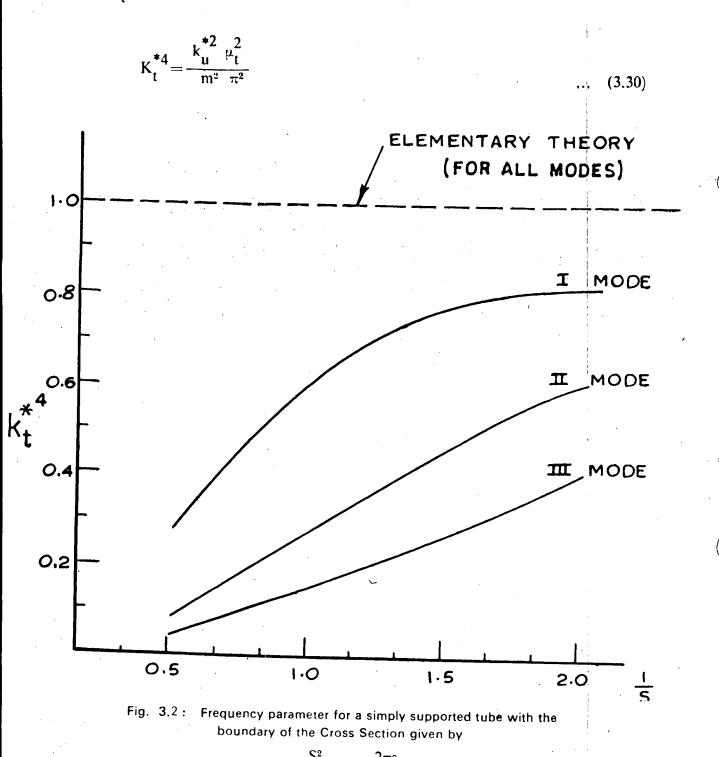
m 	Frequency pa longitudi	rameter k ^{*2} nal inertia	Modal ratio — u _{max} /w _{max} longitudinal inertia		
	Included	Ignored	Included	Ignored	
1	0.3980	0.3985	20.8744	20 9082	
2	0.7254	0.7260	22 8812	22.9337	
3	0.8559	0.8564	29.0686	29.1646	
4	0.9135	0.9138	36.3190	36.4426	
5	0.9428.	0.9431	43.9384	44.1395	

Table 3.1 - Influence of longitudinal inertia

Table 3.2 — Natural frequencies and mode shapes (n = 1)

m	First set Primarily transverse motion			Second set Primarily warping motion		
	k <mark>*2</mark> u		$-u_m/w_m$	k ^{*2} _u	u_m/w_m	
1	0.3980		20.8744	525.6650	0.0239	
2	0.7254		22.8812	288.4672	0.0275	
3.	0.8559		29.0686	244.4717	0.0171	
4	0.9135		36.3190	229.0614	0.0137	
5	0.9428		43.9385	221.9259	0.0113	

The influence of shear lag on the natural frequencies involving primarily transverse motion is shown in Fig. 3.2. These values are computed after ignoring longitudinal inertia since its effect is small. The frequency parameter used in this graph is K_r^{*4} and is given by



 $x = \frac{S^2}{4\pi^2} \sin \frac{2\pi s}{S}$

Longitudinal Inertia Ignored

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Alternately

$$K_{t}^{*4} = \left(\frac{\omega^{2} \rho L^{4} A}{E B_{xx}}\right) / m^{4} \pi^{4}$$

and the value is unity in elementary theory.

3.3 First order approximation equations

The governing equations, in this case, are obtained by putting $t_d = t_s = t$ in Eqs. (1.58 and 1.59). The equations of equilibrium are

$$S_{xx} \frac{d^{2} u}{dz^{2}} - b_{xx} \frac{d\phi_{x}}{dx} + k_{s}^{2} Au = 0 \qquad ... (3.31)$$

$$k^{2} B_{xx} \frac{d^{2} \phi_{x}}{dz^{2}} + \left(S_{xx} \frac{du}{dz} - b_{xx} \phi_{x}\right) + k_{s}^{2} B_{xx} \phi_{x} = 0$$

The boundary conditions at each end are (Eqs. 1.59)

either
$$u = 0$$
 or $S_{xx} \frac{du}{dz} - b_{xx} \phi_x = 0$... (3.32)

either
$$\phi_{\rm x} = 0$$
 or $\frac{{\rm d}\phi_{\rm x}}{{\rm d}z} = 0$

where, as defined in Eqs. (1.19), (1.55) and (1.56)

$$S_{xx} = b_{xx} = \oint \left(\frac{dx}{ds}\right)^2 t ds$$

$$B_{xx} = \oint x^2 t \, ds$$

Eqs. (3.31) may be written as

$$\frac{d^2 u}{dz^2} + k_u^2 u - \frac{d\phi_x}{dz} = 0 \qquad ... \quad (3.34)$$

and

$$\frac{d^2 \phi_x}{dz^2} + k_w^2 \phi_x - \mu_t^2 \phi_x + \mu_t^2 \frac{du}{dz} = 0 \qquad ... (3.35)$$

where $u_t^2 = \frac{S_{xx}}{k^2 B_{xx}}$

Using the following notation

$$k_{u}^{2} = \alpha_{1} K_{t}^{4} \text{ where } \alpha_{1} = 1/\mu_{t}^{2} \qquad \dots \quad (3.37)$$

$$k_{u}^{2} = \alpha_{1} K_{t}^{4} \text{ where } \alpha_{1} = R_{u}/4$$

$$k_w^2 = \beta_1 K_t^4$$
 where $\beta_1 = B_{xx}/A$

and simplifying Eqs. (3.34) and (3.35) one gets

$$\left\{\frac{d^{4}}{dz^{4}} + K_{t}^{4}\left(\alpha_{1} + \beta_{1}\right)\frac{d^{2}}{dz^{2}} - K_{t}^{4}\left(1 - K_{t}^{4}\alpha_{1}\beta_{1}\right)\right\}\left\{u \text{ or } \phi_{x}\right\} = 0$$
... (3.38)

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(2.22)

(3.33)

(3.36)

This is the same equation used by Traill-Nash and Collar⁷; the solution of this equation is discussed by them in great detail and hence will not be discussed here in detail.

The solution of the simply supported tube with the boundary of the cross section given by $x = \frac{S^2}{4\pi^2} \sin \frac{2\pi s}{S}$ by first order approximation equations is found to be the same as the exact solution for n = 1. The solution of a rectangular simply supported tube by first order approximations is summarized in section (3.4) in order to facilitate comparison with the solution by second order approximation equations.

3.4 Simply supported rectangular tube - first order approximation.

Let $-\lambda_1^2$ and λ_2^2 be the roots of the quadratic equation (Eq. (3.38))

$$\xi^{2} + K_{t}^{4}(\alpha_{1} + \beta_{1}) \xi - K_{t}^{4}(1 - K_{t}^{4}\alpha_{1}\beta_{1}) = 0 \qquad (3.39)$$

The following results are obtained by using Eqs. (3.34) and (3.35) along with the boundary conditions of the simply supported tube (Eq. (1.34)) namely,

$$u(0) = u(1) = 0$$

 $\phi'_{x}(0) = \phi'_{x}(1) = 0$... (3.40)

Appropriate expressions for u and ϕ_x can be obtained using the procedure adopted in section 2.3; satisfying the boundary conditions (3.40) the following results can be obtained

(a) $\lambda \frac{2}{2} > 0$; the secular equation is

$$\frac{\lambda_1 \lambda_2 \left(\lambda_1^2 + \lambda_2^2\right)}{\left(-\lambda_1^2 + k_u^2\right) \left(\lambda_2^2 + k_u^2\right)} \sin \lambda_1 \sinh \lambda_2 = 0$$

or the natural frequencies are given by

$$\lambda_1^2 = m^2 \pi^2$$
 $m = 1, 2, 3, ...$... (3.42)

m = 1, 2, 3, ...

the corresponding mode shapes are given by

$$u = -\frac{m \pi v_t^2}{-\lambda_1^2 + k_u^2} A'_2 \sin m \pi z$$

and

 $w = -x B_{2}' \cos m \pi z$

(3.43)

(3.41)

(3.44)

where $v_t^2 = b_{xx}/S_{xx}$

(b) $\lambda_2^2 < 0$; besides the above set of frequencies, we have another set in this

case given by

$$\lambda_2^2 = -m^2 \pi^2$$
 m = 1, 2, 3,

and the mode shapes are as given in Eqs. (3.43). It may be mentioned here that the values of k_u^2 in cases (a) and (b) are different. Besides those presented above, there is a possibility of getting an extra frequency corresponding to either $\lambda_1 = 0$ or $\lambda_2 = 0$ (Eq. 3.41) and is given by

$$k_w^2 = \mu \frac{2}{t}.$$

The natural frequencies associated with primarily transverse motion may be obtained by substituting

$$\xi = -\lambda_1^2 = -m^2 \pi^2$$

in Eq. (3.39) and ignoring longitudinal inertia, as

$$k_{u}^{2} = m^{4} \pi^{4} / (m^{2} \pi^{2} + \mu_{t}^{2})$$

which can be written as

$$K_{t}^{*4} = \frac{k_{u}^{2} \mu_{t}^{2}}{m^{4} \pi^{4}} = \mu_{t}^{2} / \left(m^{2} \pi^{2} + \mu_{t}^{2} \right) \qquad \dots \quad (3.45)$$

Some numerical results of Eq. (3.45) are discussed in section (3.8)

3.5 Second order approximation equations

The governing equations in this case are obtained by putting $t_d = t_s = t$ in Eq. (1.84) The equations of equilibrium are

$$S_{xx} \frac{d^{2}u}{dz^{2}} - b_{xx} \frac{d\phi_{x}}{dz} - \iota_{xx} \frac{d\psi_{x}}{dz} + k_{s}^{2} Au = 0$$

$$k^{2} \left(B_{xx} \frac{d^{2}\phi_{x}}{dz^{2}} + \bar{\iota}_{xx} \frac{d^{2}\Psi_{x}}{dz^{2}} \right) + \left(S_{xx} \frac{du}{dz} - b_{xx} \phi_{x} - L_{xx}\Psi_{x} \right)$$

$$+ k_{s}^{2} \left(B_{xx} \phi_{x} + \bar{\iota}_{xx} \Psi_{x} \right) = 0$$

$$k^{2} \left(\bar{\iota}_{xx} \frac{d^{2}\phi_{x}}{dz^{2}} + \bar{L}_{xx} \frac{d^{2}\psi_{x}}{dz^{2}} \right) + \left(\iota_{xx} \frac{du}{dz} - L_{xx}\phi_{x} - \mathcal{L}_{xx}\psi_{x} \right)$$

$$+ k_{s}^{2} \left(\bar{\iota}_{xx} \frac{d^{2}\phi_{x}}{dz^{2}} + \bar{L}_{xx} \frac{d^{2}\psi_{x}}{dz^{2}} \right) + \left(\iota_{xx} \frac{du}{dz} - L_{xx}\phi_{x} - \mathcal{L}_{xx}\psi_{x} \right)$$

$$+ k_{s}^{2} \left(\bar{\iota}_{xx} \phi_{x} + \bar{\iota}_{xx} \psi_{x} \right) = 0 \qquad ... (3.46)$$

August 1968] GENERAL THEORY OF VIBRATIONS OF CYLINDRICAL TUBES and the boundary conditions at each end are (Eqs. (1.85)) either u = 0 or $S_{xx} \frac{du}{dz} - b_{xx} \phi_x - b_{xx} \psi_x = 0$ either $\phi_x = 0$ or $B_{xx} \frac{d\phi_x}{dz} + \bar{b}_{xx} \frac{d\psi_x}{dz} = 0$ either $\psi_x = 0$ or $\overline{L}_{xx} \frac{d0_x}{dz} + \overline{L}_{xx} \frac{d\psi_x}{dz} = 0$ (3.47)

In obtaining above, the following relation has been used

$$w = -x \phi_{x} - \overline{w}_{2}, x \psi_{x}$$

$$\overline{w}_{2}, x = -\iint x \, ds \, ds$$

$$\dots \quad (3.48)$$

$$\dots \quad (3.49)$$

where

The two constants of integration in the above integral have to be evaluated using the condition of continuity of $\frac{d\overline{w}_{2,x}}{ds}$ and the condition of zero net axial force. Various cross-sectional constants involved in Eqs. (3.46) and (3.47) are given below (Eqs. (1.19). (1.55), (1.56), (1.82) and (1.83))

$$S_{xx} = b_{xx} = \oint \left(\frac{dx}{ds}\right)^2 t \, ds$$

$$L_{xx} = L_{xx} = \oint \frac{dx}{ds} \quad \frac{d\overline{w}_{2,x}}{ds} t \, ds$$

$$\pounds_{xx} = \oint \left(\frac{d\overline{w}_{2,x}}{ds}\right) t \, ds$$

$$B_{xx} = \oint x^2 t \, ds$$

$$\overline{L}_{xx} = \oint x \, \overline{w}_{2,x} t \, ds$$

$$\overline{L}_{xx} = \oint (\overline{w}_{2,x})^2 t \, ds$$

36 Cross-sectional constants of a rectangular tube-second approximation equations.

Since the tube under consideration is doubly symmetric as shown in Fig. (3.3) and since we are considering vibration in the plane of XOZ, warp is symmetric about FB (Fig. 3.3) and antisymmetric about OD. Therefore, it is sufficient to consider the region OAB for evaluation of \overline{w}_2 , x and the distribution of \overline{w}_2 , x in other regions may be obtained considering the above features.

x in the region OAB is given by

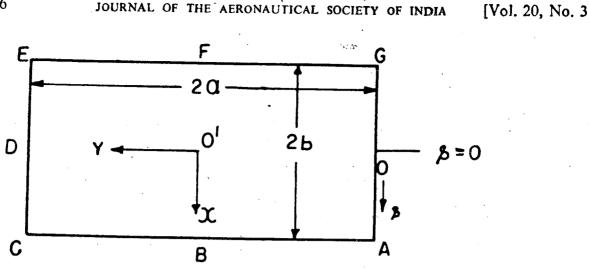
$$\mathbf{x} = \mathbf{s} \quad 0 \leq \mathbf{s} \leq \mathbf{b}$$

 $= b \quad b \leq s \leq a + b$

(3.50)

... (3.51)

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ig. 3,3 : A Rectangular Tuba.

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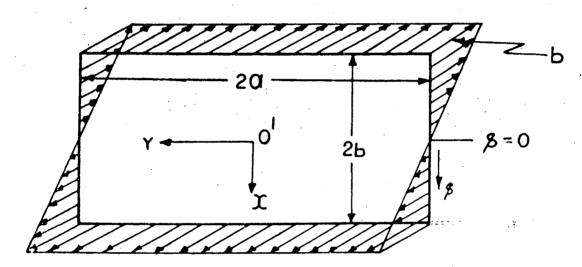


Fig. 3.4 : x = x (s) in a Rectangular Tube.

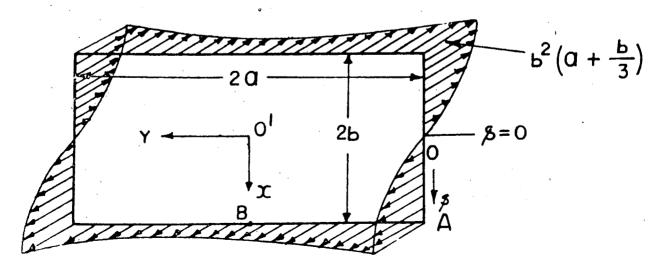


Fig. 3.5 : $\overline{\mathbf{w}}_{2\mathbf{x}}$ in a Rectangular Tube.

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(3.52)

... (3.54)

 \overline{w}_{2} , x is obtained from the relation given in Eq. (3.49). From symmetry and antisymmetry, we have

$$(\overline{w}_{2,x})_{x=0} = 0$$
$$\left(\frac{d\overline{w}_{2,x}}{ds}\right)_{s=a+b} = 0$$

and

 $\bar{w}_{2, x}$ and $\frac{d\bar{w}_{2, x}}{ds}$ have to be continuous at A.

Using Eqs. (3.51) in (3.49) and satisfying conditions (3.52), we obtain

$$\overline{w}_{2,x} = -\frac{s^{3}}{6} + (b^{2}/2 + ab) \text{ s in } 0 \le s \le b$$

= $-\frac{b(s-b)^{2}}{2} + ab (s-a) + b^{2} (a+b/3) \text{ in } b \le s \le a + b$
.... (3.53)

The distributions of x and $\overline{w}_{2,x}$ are shown in Figs. (3.4) and (3.5).

Using Eqs. (3.51) and (3.53) in Eqs. (3.50), the cross-sectional constants are evaluated as

$$S_{xx} = b_{xx} = 4bt$$

$$L_{xx} = L_{xx} = B_{xx} = 4 b^{2} t (a+b/3)$$

$$\mathcal{L}_{xx} = \overline{L}_{xx} = \frac{4b^{2}t}{15} (5a^{3} + 15a^{2}b + 25ab^{2} + 2b^{3})$$

$$\overline{L}_{xx} = \frac{4b^{2}t}{630} (84a^{5} + 420a^{4}b + 770a^{3}b^{2} + 630 a^{2}b^{3} + 238 ab^{4} + 34 b^{5})$$

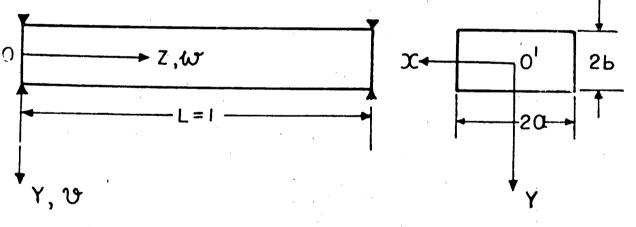


Fig. 3.6: A simply supported Rectangular Tube.

3.7 Simply supported rectangular tube-second order approximation equations

The equations of equilibrium are given by Eqs. (3.46); while the boundary conditions are Eqs. (3.47).

$$u(0) = u(1) = 0 \qquad ... (3.55)$$

$$\phi'_{x}(0) = \phi'_{x}(1) = 0 \qquad ... (3.56)$$

$$\Psi'_{x}(0) = \Psi'_{x}(1) = 0 \qquad ... (3.57)$$

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Eqs. (3.46) may be written as

$$u'' - \phi'_{x} - v_{1}^{2} \quad \Psi'_{x} + k_{u}^{2} u = 0 \qquad (3.58)$$

$$\phi''_{x} + \left(k_{w}^{2} - \Gamma_{1}^{2} \right) \phi_{x} + u_{1}^{2} \left\{ \Psi''_{x} + \left(k_{w}^{2} - \Gamma_{3}^{2} \right) \Psi_{x} \right\} + \Gamma_{1}^{2} u' = 0 \qquad (3.59)$$

$$\phi''_{x} + \left(k_{w}^{2} - \Gamma_{2}^{2} \right) \phi_{x} + u_{2}^{2} \left\{ \Psi''_{x} + \left(k_{w}^{2} - \Gamma_{4}^{2} \right) \Psi_{x} \right\} + \Gamma_{2}^{2} u' = 0 \qquad (3.60)$$

where

Combining Eqs. (3.58), (3.59) and (3.60), we have

where

D denotes d/dz

and
$$\lambda_{1}^{2} = -\frac{\mu_{1}^{2} \Gamma_{3}^{2} - \mu_{2}^{2} \Gamma_{4}^{2} + \nu_{1}^{2} (\Gamma_{2}^{2} - \Gamma_{1}^{2})}{\mu_{2}^{2} - \mu_{1}^{2}}$$

 $\xi_{1}^{2} = \frac{\mu_{1}^{2} (\Gamma_{2}^{2} + \Gamma_{3}^{2}) - \mu_{2}^{2} (\Gamma_{1}^{2} + \Gamma_{4}^{2})}{\mu_{1}^{2} \Gamma_{3}^{2} - \mu_{2}^{2} \Gamma_{4}^{2} + \nu_{1}^{2} (\Gamma_{2}^{2} - \Gamma_{1}^{2})}$... (3.63)
 $\xi_{2}^{2} = \frac{\mu_{2}^{2} \Gamma_{1}^{2} \Gamma_{4}^{2} - \mu_{1}^{2} \Gamma_{2}^{2} \Gamma_{3}^{2}}{\mu_{2}^{2} - \mu_{1}^{2}}$

From Eq. (3.62), the expressions for u, ϕ_x and Ψ_x are,

$$u = A_{1} \sin \lambda_{1} z + A_{2} \cos \lambda_{1} z + A_{3} \sinh \lambda_{2} z + A_{4} \cosh \lambda_{2} z + A_{5} \sinh \lambda_{3} z + A_{5} \cosh \lambda_{3} z \psi_{x} = A_{1}^{1} \sin \lambda z + A_{2}^{1} \cos \lambda_{1} z + A_{3}^{1} \sinh \lambda_{2} z + A_{4}^{1} \cosh \lambda_{2} z + A_{5}^{1} \sinh \lambda_{3} z + A_{6}^{1} \cosh \lambda_{3} z$$

$$\psi_{\mathbf{x}} = \mathbf{A}_{1}^{"} \sin \lambda_{1} \mathbf{z} - \mathbf{A}_{2}^{"} \cos \lambda_{1} \mathbf{z} - \mathbf{A}_{3}^{"} \sinh \lambda_{2} \mathbf{z} + \mathbf{A}_{4}^{"} \cosh \lambda_{2} \mathbf{z}$$

+ $A_5^* \sinh \lambda_3 z - A_6^* \cosh \lambda_3 z$

... (3.64)

where

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and

$$C_{1} = k_{u}^{2} + 2k_{w}^{2} - \lambda_{t}^{2}$$

$$C_{2} = k_{w}^{2} \left(2k_{u}^{2} + k_{w}^{2} \right) - \lambda_{t}^{2} \left(k_{w}^{2} + \zeta_{1}^{2} k_{u}^{2} \right) \qquad \dots \quad (3.66)$$

$$C_{3} = k_{u}^{2} \left\{ k_{w}^{2} \left(k_{w}^{2} - \lambda_{t}^{2} \zeta_{1}^{2} \right) + \zeta_{2}^{2} \right\}$$

In writing Eqs (3.64), it is assumed that λ_1^2 , λ_2^2 and λ_3^2

are positive. However, situations may exist in which either of λ_2^2 and λ_3^2

or both may be negative, but such cases are expected to represent primarily warping motion. As the present interest lies in assessing the influence of shear lag on the natural frequencies associated with primarily transverse motion, the case of negative λ_2^2 and λ_3^2 is excluded from the present discussion.

It is obvious that all the arbitrary constants in Eqs. (3.64) are not independent for they have to satisfy any two of Eqs. (3.58), (3.59) and (3.60). Substitution of Eqs. (3.64) in Eqs. (3.58) and (3.59) yields the following sets of relationships respectively.

$$\left(-\lambda_{1}^{2} + k_{u}^{2} \right) A_{1} + \lambda_{1} A_{2}^{1} + \lambda_{1} \upsilon_{1}^{2} A_{2}^{*} = 0 \left(-\lambda_{1}^{2} + k_{u}^{2} \right) A_{2} - \lambda_{1} A_{1}^{1} + \lambda_{1} \upsilon_{1}^{2} A_{1}^{*} = 0 \left(\lambda_{2}^{2} + k_{u}^{2} \right) A_{3} - \lambda_{2} A_{4}^{1} - \lambda_{2} \upsilon_{1}^{2} A_{4}^{*} = 0 \left(\lambda_{2}^{2} + k_{u}^{2} \right) A_{4} - \lambda_{2} A_{3}^{1} - \lambda_{2} \upsilon_{1}^{2} A_{3}^{*} = 0 \left(\lambda_{2}^{2} + k_{u}^{2} \right) A_{4} - \lambda_{2} A_{3}^{1} - \lambda_{2} \upsilon_{1}^{2} A_{3}^{*} = 0 \left(\lambda_{3}^{2} + k_{u}^{2} \right) A_{5} - \lambda_{3} A_{6}^{1} - \lambda_{3} \upsilon_{1}^{2} A_{6}^{*} = 0 \left(\lambda_{3}^{2} + k_{u}^{2} \right) A_{6} - \lambda_{3} A_{5}^{1} - \lambda_{3} \upsilon_{1}^{2} A_{5}^{*} = 0$$

and

$$\Gamma_{1}^{2} \lambda_{1} A_{1} + \left(-\lambda_{1}^{2} + k_{w}^{2} - \Gamma_{1}^{2}\right) A_{2}^{1} + \mu_{1}^{2} \left(-\lambda_{1}^{2} + k_{w}^{2} - \Gamma_{3}^{2}\right) A_{2}^{*} = 0$$

$$- \Gamma_{1}^{2} \lambda_{1} A_{2} + \left(-\lambda_{1}^{2} + k_{w}^{2} - \Gamma_{1}^{2}\right) A_{1}^{1} + \mu_{1}^{2} \left(-\lambda_{1}^{2} + k_{w}^{2} - \Gamma_{3}^{2}\right) A_{1}^{*} = 0$$

$$\Gamma_{1}^{2} \lambda_{2} A_{3} + \left(-\lambda_{2}^{2} + k_{w}^{2} - \Gamma_{1}^{2}\right) A_{4}^{1} + \mu_{1}^{2} \left(-\lambda_{2}^{2} + k_{w}^{2} - \Gamma_{3}^{2}\right) A_{4}^{*} = 0$$

$$\Gamma_{1}^{2} \lambda_{2} A_{3} + \left(-\lambda_{2}^{2} + k_{w}^{2} - \Gamma_{1}^{2}\right) A_{4}^{1} + \mu_{1}^{2} \left(-\lambda_{2}^{2} + k_{w}^{2} - \Gamma_{3}^{2}\right) A_{4}^{*} = 0$$

$$\Gamma_{1}^{2} \lambda_{2} A_{4} + \left(-\lambda_{2}^{2} + k_{w}^{2} - \Gamma_{1}^{2}\right) A_{3}^{1} + \mu_{1}^{2} \left(-\lambda_{2}^{2} + k_{w}^{2} - \Gamma_{3}^{2}\right) A_{3}^{*} = 0$$

$$\Gamma_{1}^{2} \lambda_{3} A_{5} + \left(\lambda_{3}^{2} + k_{w}^{2} - \Gamma_{1}^{2}\right) A_{6}^{1} + \mu_{1}^{2} \left(\lambda_{3}^{2} + k_{w}^{2} - \Gamma_{3}^{2}\right) A_{6}^{*} = 0$$

$$\Gamma_{1}^{2} \lambda_{3} A_{6} + \left(\lambda_{3}^{2} + k_{w}^{2} - \Gamma_{1}^{2}\right) A_{5}^{1} + \mu_{1}^{2} \left(\lambda_{3}^{2} + k_{w}^{2} - \Gamma_{3}^{2}\right) A_{5}^{*} = 0$$

... (3.68)

Satisfaction of the boundary conditions at z = 0, namely $u(0) \phi'_{x}(0)$

and

$$\Psi'_{X}(0) = 0 \text{ yield} A_{2} + A_{4} + A_{6} = 0 \lambda_{1} A_{1}^{1} + \lambda_{2} A_{3}^{1} + \lambda_{3} A_{5}^{1} = 0 \qquad \dots (3.69) \lambda_{1} A_{1}'' + \lambda_{2} A_{3}'' + \lambda_{3} A_{5}'' = 0$$

It can be shown from Eqs. (3.69) with the help of Eqs. (3.67) and (3.68) that

$$A_2 = A_4 = A_6 = A_1^1 = A_3^1 = A_5^1 = A_1'' = A_3'' = A_5'' = 0$$

... (3.70)

Satisfying the boundary conditions at z = 1, namely u (1) = 0, $\phi_x^1(1) = 0$ and $\Psi_x^1 = 0$, and using Eqs. (3.70) one obtains

$$A_{1} \sin \lambda_{1} + A_{3} \sinh \lambda_{2} + A_{5} \sinh \lambda_{3} = 0$$

- $\lambda_{1}A_{2}^{1} \sin \lambda_{1} + \lambda_{2}A_{4}^{1} \sinh \lambda_{2} + \lambda_{3}A_{6}^{1} \sinh \lambda_{3} = 0$
- $\lambda_{1}A_{2}^{\prime\prime} \sin \lambda_{1} + \lambda_{2}A_{4}^{\prime\prime} \sinh \lambda_{2} + \lambda_{3}A_{6}^{\prime} \sinh \lambda_{3} = 0$
(3.71)

Using Eqs. (3.67), (3.68) and (3.71) it can be shown that for nontrivial solution

or

$$\lambda_1 = m \pi \quad m = 1, 2, 3 \dots 00$$
 ... (3.73)

Since our interest at this stage is confined only to primarily transverse vibrations, the corresponding mode shapes are

$$u = A_1 \sin m \pi z$$
 ... (3.74)

$$w = -x\phi_{x} - \bar{w}_{2,x} \Psi_{x} = -x A_{2}^{1} + \bar{w}_{2,x} A_{2}^{*} \cos m \pi z$$

where A_2^1 and A_2^r are obtained in terms of A_1 by solving the two first equations in Eqs. (3.67) and (3.68).

Neglecting longitudinal inertia a simple expression for natural frequency can be obtained; substituting

$$\xi = -\lambda_1^2 = -m^2 \pi^2 \qquad ... \quad (3.75)$$

in Eq. (3.65); ignoring longitudinal inertia $(k_w^2 = 0)$, and simplifying

$$k_{u}^{2} = \frac{m^{6} \pi^{6} + \lambda_{t}^{2} m^{4} \pi^{4}}{m^{4} \pi^{4} + m^{2} \pi^{2} \lambda_{t}^{2} \zeta_{1}^{2} + \zeta_{2}^{2}}$$

which can also be written as

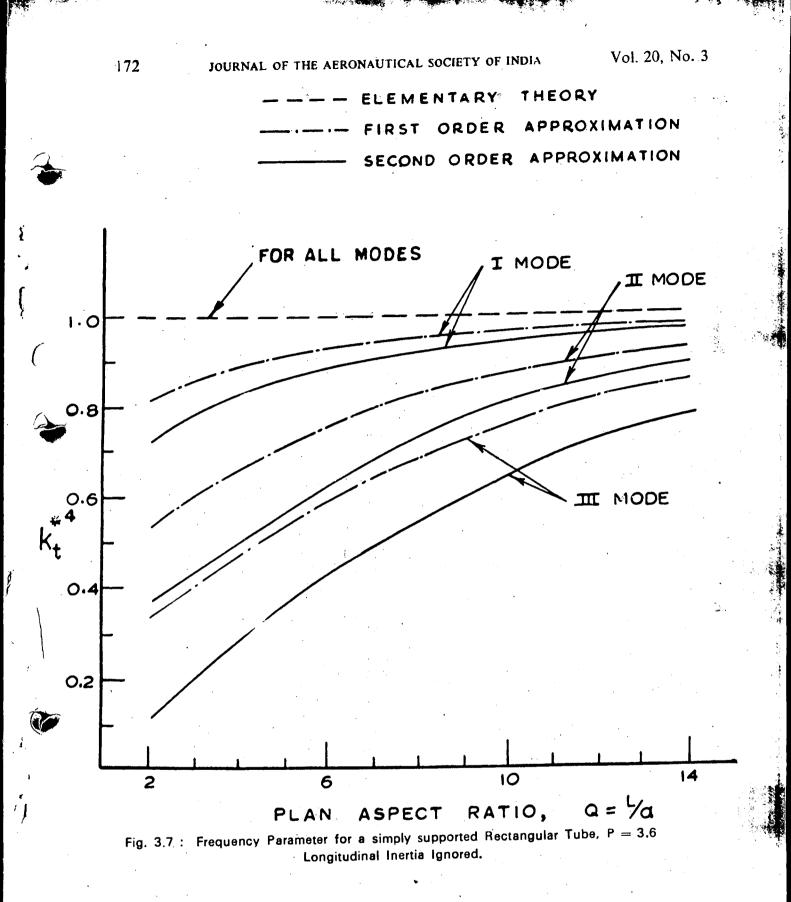
$$K_{t}^{*} = \frac{k_{u}^{2} \mu_{t}^{2}}{m^{4} \pi^{4}} = \frac{\mu_{t}^{2} \left(m^{2} \pi^{2} + \lambda_{t}^{2}\right)}{m^{4} \pi^{4} + m^{2} \pi^{2} \lambda_{t}^{2} \zeta_{1}^{2} + \zeta_{2}^{2}} \dots (3.76)$$

The values of K_t^{*4} , obtained by using Eqs. (3.45) and (3.76) and presented in Table 3.3, denote the results obtained by using first and second order approximations respectively.

Р	Q	Order of approximation	Fundamental	Second mode	Third mode
3.6	2	First	0.819	0.531	0.335
		Second	0.730	0.367	0.115
	6	First	0.926	0.759	0.583
		Second	0.883	0.638	0.424
	10	First	0.966	0.876	0.759
		Second	0.946	0.810	0.648
	14	First	0.981	0.926	0.848
		Second	0.969	0.885	0.770

Table 3.3—Influence of shear lag on Kt

The results of the first approximation can be obtained from the equations presented in Ref. (7) also. The results of Table (3.3) are presented in a graphical form in Fig. 3.7. This reveals that the effect of shear lag is considerable particulary at short aspect ratios and this effect is more significant for higher modes. Although there



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are no exact results to provide comparison for the results presented in Table 3.3, the trends are in agreement with the results of Ref. (13) which considers a free-free tube in transverse vibration.

39 Conclusions

In this chapter problems of flexural vibrations of doubly symmetric unstiffened closed tubes have been considered. The exact solution of a simply supported tube with the cross section given by $x = \frac{S^2}{4\pi^2} \sin \frac{2\pi s}{s}$ has been presented. This reveals that the frequency spectrum in this case includes a doubly infinite set of frequencies besides the well known infinite set of frequencies involving primarily transverse motion. One infinite set of this additional doubly infinite set involves large warping motion associated with small transverse motion while the remaining pertain to pure warping motions.

The numerical results reveal that the influence of shear lag is considerable for short aspect ratios and increases for higher modes. From this it can be inferred that first order approximation equations are not adequate; it is essential to consider at least second order approximation equations to get reasonably good results.

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